

SGO LOWEST

Cost savings through payload reduction in a linear configuration. John Baker(GSFC) - for SGO Core Team

Workshop for Gravitational-Wave Mission Architectural Concepts 20-21 Dec 2011

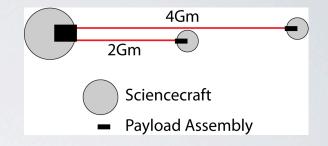
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RATIONALE

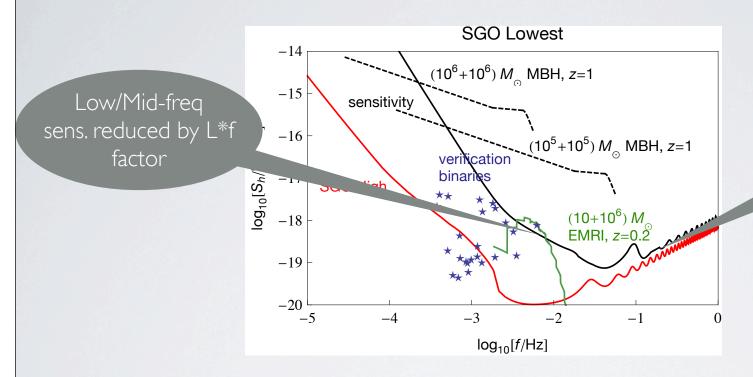
- Objective: Conceive lowest-cost concept preserving some of LISA science
- Cannot eliminate a spacecraft or much of payload subsystems
- Save by: aggressively reducing payload multiplicity
- Leads to: linear 'three daughter''concept
- 3 S/C kept maximally similar to minimize nonrecurring expenses.

LINEAR GW MEASUREMENT

- How is laser frequency noise dealt with?
- Laser frequency noise is canceled using Time-Delay-Interferometry (TDI) exactly as with LISA.
- Like S/C Doppler tracking meas. of GW using two S/C to cancel laser noise. Need 3 separated S/C.
- Measure temporal GW variations:
 - High-freq: sensitivity compares with LISA
 - Low-freq: limited sensitivity.



SCIENCE



Box 1. The black curve shows SGO Lowest's rms strain noise, in units of $Hz^{-1/2}$. Roughly speaking

L High-freq sens. re, comparable to LISA stree

binaries"). The two dashed black curves and the dashed green curve represent sources (two SMBH binaries, and an EMRI, respectively) whose frequency evolves upward significantly during the observation.

- Dramatically lower sensitivity below 30 mHz
- Only minimal LISA science is achievable 2/10

SCIENCE COMPARISON

Comparison of Science Performance for different versions of SGO					
Concept	SGO High	SGO Mid	SGO Low	SGO Lowest	
Nominal Lifetime	5 yrs	2 yrs	2 yrs	2 yrs	
MBH mergers					
Total # Detections	$70 \sim 150$	$25 \sim 35$	$25 \sim 35$	~ 4	
Median Redshift	$\tilde{z} \sim 5$	$\tilde{z} \sim 5$	$\tilde{z} \sim 5$	$\tilde{z} \sim 4$	
Mass Precision @ $z = \tilde{z}$	$\frac{\sigma_M}{M} \sim 0.2\%$	$\frac{\sigma_M}{M} \sim 1\%$	$\frac{\sigma_M}{M} \sim 1\%$	$\sim 3\%$	
Spin Accuracy $@ z = \tilde{z}$	$\sigma\chi\sim 0.3\%$	$\sigma\chi\sim 2\%$	$\sigma\chi\sim 3\%$	-	
Distance Accuracy @ $z = \tilde{z}$	$\frac{\sigma_{D_L}}{D_L} \sim 3\% \text{ (WL)}$	$\frac{\sigma_{D_L}}{D_L} \sim 3\% \text{ (WL)}$	$\frac{\sigma_{D_L}}{D_L} \sim 20\%$	-	
Sky Localization @ $z = \tilde{z}$	$\sim 1 \ \mathrm{deg}^2$	$\sim 1 \ \mathrm{deg}^2$	$\gtrsim 100 \ \mathrm{deg}^2$	-	
# Detections @ $z < 2$	~ 7	$1 \sim 2$	$1 \sim 2$	< 1	
Mass Precision @ $z = 1$	$\frac{\sigma_M}{M} \lesssim 0.1\%$	$\frac{\sigma_M}{M} \lesssim 0.1\%$	$\frac{\sigma_M}{M} \lesssim 0.3\%$	-	
Spin Accuracy $@ z = 1$	$\sigma\chi \lesssim 0.1\%$	$\sigma\chi \lesssim 0.1\%$	$\sigma\chi \lesssim 1\%$	-	
Sky Localization $@z = 1$	$\lesssim 0.1 \ { m deg}^2$	$\lesssim 0.1 \ \mathrm{deg^2}$	$\lesssim 10 \ { m deg}^2$	-	
EMRIs					
# Detections	$40 \sim 4000$, to $z \sim 1.0$	$2 \sim 200$, to $z \sim 0.2$	$\lesssim 40$, to $z \sim 0.15$	0	
Mass Accuracy	$\frac{\sigma_M}{M} \sim 0.01\%$	$\frac{\sigma_M}{M} \sim 0.01\%$	$\frac{\sigma_M}{M} \sim 0.01\%$	-	
MBH Spin Accuracy	$\sigma\chi\sim 0.01\%$	$\sigma\chi\sim 0.01\%$	$\sigma\chi\sim 0.01\%$		
Compact Binaries					
# Verification binaries	10	8	7	0	
# Resolvable binaries	$\sim 20,000$	$\sim 4,000$	$\sim 2,000$	~ 100	
Discovery Space					
Detects early-universe Ω_{gw}	$\gtrsim 10^{-10}$	$\gtrsim 10^{-9}$	-	-	
Can Detect+Verify Bursts?	\checkmark	\checkmark	-	-	

RISK

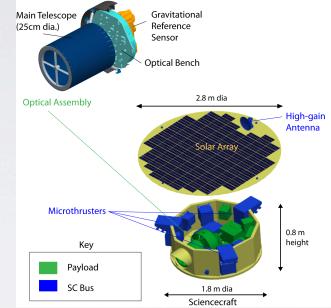
- Elimination of redundancy (vs. LISA)
 -single-string design
 -Only 4 links, 1 IFO
- Stringent orbit requirements

 stationkeeping likely requires mN range for µN thrusters
 trajectory risk without P/M
- Some additional complications

 -double infield guiding
 -double heterodyne separation of two far S/C signals
- Plausible risk that no MBHs are detected

COST SAVINGS

SGO High estimate	1.66
Launch vehicle savings	-0.01
Optical assembly count reduction	-0.13
Payload mass or redundancy reduction	-0.11
Mission duration reduction	-0.11
Propulsion module elimination	-0.11
SGO Lowest total	\$1.19B



- Savings:
 - Roughly half of payload eliminated (one telescope/proof-mass per S/C)
 - Simple circular drift-away orbits may also allow elimination of prop. module
 - Shortened operational phase (strictly ~2 yrs)
- Cost model based on LISA; accounts for NRE, replication learning curve, parameterized mass-scaling of payload+SC
- Modest additional savings (~\$0.1B) might be achieved by GRS elimination as in McKenzie concept

SUMMARY

- Linear concept seems close to lowest possible cost concept
- Only minimal LISA science is achievable:
- ...with significant additional risks
- Cost \$1.1-1.2B: saves 35% of SGO-high (i.e. US LISA) costs
- A comparably estimated lower-cost mission seems unlikely.